<b>X</b>			3.a.2.	the organisation must implement and maintain a management system to ensure compliance with these essential requirements for airworthiness, and aim for continuous improvement of this system;			
			3.a.3.	the organisation must establish arrangements with other relevant organisations, as necessary, to ensure continuing compliance with these essential requirements for airworthiness;			
			3.a.4.	the organisation must establish an occurrence reporting and/or handling system, which must be used by the management system under point 3.a.2 and the arrangements under point 3.a.3, in order to contribute to the aim of continuous improvement of the safety of products.			
			3.b.	In the case of maintenance training organisations, the conditions under points 3.a.3 and 3.a.4 do not apply.			
	ANNEX II Aircraft referred to in Article 4(4)						
	ANNEX II		Article 4 categor	Article $4(1)$ , $(2)$ and $(3)$ do not apply to aircraft falling in one or more of the categories set out below:			
			(a) hi	(a) historic aircraft meeting the criteria below:			
			(i)	(i) non-complex aircraft whose:			
			- i	- initial design was established before 1 January 1955,			
			- 1	and production has been stopped before 1 January 1975. or			
			(ii	i) aircraft having a clear historical relevance, related to:			



- a participation in a noteworthy historical event.
or
- a major step in the development of aviation.
or
<ul> <li>a major role played into the armed forces of a Member State;</li> </ul>
(b) aircraft specifically designed or modified for research, experimental or
scientific purposes, and likely to be produced in very limited numbers;
(a) since $f$ of which at least $\Gamma$ 1.0( is built by an exception of a new multiplication of $\Gamma$
(c) aircraft of which at least 51 % is built by an amateur, or a non-profit making association of amateurs for their own purposes and without any
commercial objective;
(d) aircraft that have been in the service of military forces, unless the aircraft
is of a type for which a design standard has been adopted by the Agency.
(e) aeroplanes, helicopters and powered parachutes having no more than two
seats, a maximum take-off mass (MTOM), as recorded by the Member
(i) 300 kg for a land plane/helicopter, single-seater.
or
(ii) 450 kg for a land plane/helicopter, two-seater.
or
(iii) 330 kg for an amphibian or floatplane/helicopter single-seater;
or

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	<ul> <li>(iv) 495 kg for an amphibian or floatplane/helicopter two-seater, provided that, where operating both as a floatplane/helicopter and as a land plane/helicopter, it falls below both MTOM limits, as appropriate.</li> <li>(v) 472,5 kg for a land plane, two-seater equipped with an airframe mounted total recovery parachute system.</li> </ul>	
	(vi) 315 kg for a land plane single-seater equipped with an airframe mounted total recovery parachute system;	
	<b>and</b> , for aeroplanes, having the stall speed or the minimum steady flight speed in landing configuration not exceeding 35 knots calibrated air speed (CAS);	
	(f) Single and two-seater gyroplanes with a maximum take off mass not exceeding 560 kg.	
	(g) Gliders with a maximum empty mass, of no more than 80 kg when single- seater or 100 kg when two-seater, including those which are foot launched.	
	(h) Replicas of aircraft meeting the criteria of (a) or (d) above, for which the structural design is similar to the original aircraft.	
	(i) Unmanned aircraft with an operating mass of no more than 150 kg.	
	(j) any other aircraft which has a maximum empty mass, including fuel, of no more than 70 kg.	